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OPEX-II, A RADIATION SHIELD OPTIMIZATION CODE

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ABSTRACT

A radiation shield optimization procedure based on the computer code, OPEX-II, is described. The OPEX-II code, based on an earlier, steepest-descent-method code OPEX, has been recoded to improve coding, simplify data input, use spherical geometry, and alter the dose-thickness relation when a layer has been removed. A complete description of how to obtain the necessary input data for OPEX-II from other transport calculations is given. Data input instructions, FORTRAN IV code listing, and a sample problem optimizing a seven-layer shield of tungsten and lithium hydride for a space power reactor are given.

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SUMMARY

A radiation shield optimization procedure based on the computer code, OPEX-II, is described. The OPEX-II code, based on an earlier, steepest-descent-method code OPEN, has been recoded to improve coding, simplify data input, use spherical geometry, and alter the dose-thickness relation when a layer has been removed. A complete description of how to obtain the necessary input data for OPEX-II from other transport calculations is given. Data input instructions, FORTRAN IV code listing, and a sample problem optimizing a seven layer shield of tungsten and lithium hydride for a space power reactor are given.

INTRODUCTION

The radiation shield designer is faced with the task of selecting shield materials and material arrangements which will not only provide adequate protection against radiation but will also minimize shield weight, cost, or space. If components of the total dose are independent of one another (i.e., primary gammas, and fission neutrons), and the geometry is simple, analytic expressions for the radiation dose may be defined and a closed form solution for minimum weight may be obtained. Reference 1 reviews such cases.

When radiation groups are not independent, which is generally the case for secondary gammas generated by neutron absorptions and inelastic scatters throughout the shield, closed form solutions are no longer attainable so numerical iterative methods must be employed. Two of these numerical, iterative optimization techniques are (1) the method of Lagrange multipliers (as applied to the shield weight optimization problem in ref. 2) and (2) the method of steepest descent (as applied to the shield weight optimization problem in ref. 3).

In both methods, an empirical analytical expression (hereinafter called the dosethickness relation) is assumed which relates the radiation dose at some reference deexpression are obtained by fitting them to some accurate detailed radiation transport calculations of dose for a given base configuration and perturbations of that configuration. The geometry and thicknesses of material determine weight and derivative of weight with respect to thickness. With first derivative of weight and dose with respect to thickness as determined from the dose-thickness relation, the optimization procedure alters the base configuration to obtain a set of thicknesses corresponding to a minimum weight configuration (or at least a local minimum) for some dose constraint. Optimization codes do not select materials but merely alter initial configurations. Optimization codes may eliminate layers but they never add layers. The optimum weight estimate is only as good as the parameters which describe changes of dose with thickness. A final, detailed proof calculation is necessary to confirm the predictions of the optimization code.

The steepest-descent method of reference 3 was incorporated in a rudimentary computer program called OPEX (ref. 4) but was limited to slab geometry. This report describes a revision of OPEX, called OPEX-II, and how it is applied to a radiation shield optimization. The basic steepest-descent method of reference 3 has been maintained, but the code OPEX has been completely rewritten to improve coding, simplify data input, use spherical geometry, expand the output, and alter the dose-thickness relation when a layer has been removed by the optimization code. In this report, a complete description of how to obtain the necessary input data for OPEX-II from other transport calculations is given. Data input instructions, FORTRAN IV code listing, and a sample problem optimizing a seven-layer shield of tungsten and lithium hydride for a space power reactor are given.

DOSE-THICKNESS RELATION

The total radiation dose rate D at some reference point in space is defined, for purposes of the optimization procedure, to be

$$D = \sum_{i}^{IMAX} D_{i}$$

where

D total dose rate

D_i ith component of total dose rate (e.g., dose due to capture gammas from first shield layer or dose due to inelastic gammas from last shield layer, etc.)

IMAX number of components of dose

Each dose component D_i is further assumed to be of the form

$$D_{i} = C_{i} \exp \left(-\sum_{j=1}^{NREG} \mu_{ij} t_{j} \right)$$
 (2)

where

C_i fitted parameter

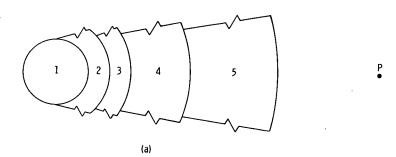
t_j thickness of jth region

 μ_{ij} 'attenuation coefficient' which describes effect of change in thickness t_j on D_i

NREG number of regions present

The ith dose component need not be associated with the jth region. There may be more than one contribution to the total dose from a given layer. Other layers may contribute negligible secondary gamma dose and their dose contribution is omitted. Generally because of differences in formation rate and gamma attenuation, a given high-Z shield region will have one dose component due to capture gammas and a second due to inelastic gammas. The core will have primary neutrons and primary gammas. A hydrogeneous layer may have negligible dose.

For example, consider the spherical reactor-shield described in the following sketch and table with a dose constraint at point P:



Region j	Description	Radiation dose com- ponent i	Description
1	Reactor core	1 2	Core neutron Core gamma
2	Reflector	3 4	Capture gamma Inelastic gamma
3	Lithium hydride		Negligible dose
4	Tungsten shield	5 6	Capture gamma Inelastic gamma
5	Lithium hydride		Negligible dose

In this example, $\mu_{2,4}$, $\mu_{3,4}$, and $\mu_{4,4}$ represent the attenuation of gamma rays by the tungsten layer for the various gamma sources in the core and layers between the core and the tungsten. The coefficient $\mu_{1,4}$ represents the attenuation by the tungsten of neutron dose due to neutrons born in the core. In contrast, $\mu_{5,3}$ represents the attenuation by lithium hydride (layer 3) of neutrons which give rise to secondary gamma sources in the tungsten (layer 4).

For coefficients such as $\mu_{5,4}$, describing the effect of a thickness on its own source strength, equation (2) becomes inadequate particularly when the thickness of the layer goes to zero. OPEX-II does, however, set $C_i = 0$ if a particular layer is eliminated by the code. At this point, however, the attenuation coefficients μ_{ij} should be recalculated for the new configuration. Regions of origin of the various dose components are required data input to the code to facilitate this operation.

FITTING PARAMETERS TO DOSE-THICKNESS RELATION

The coefficients of equation (2) are obtained as precisely as possible by performing a series of transport calculations. A starting, base configuration is selected, preferably as close to an optimum configuration as possible based on past experience. With the transport calculation, such as a discrete ordinates calculation, the individual dose components D_i are evaluated. Each layer of the base configuration to be altered by the optimization code is then systematically increased by a nominal amount (say, 1 cm) and the dose components are reevaluated with the transport code. From equation (2), then,

$$D_{\mathbf{i}}(t_1, t_2, \dots, t_j, \dots, t_{NREG}) = C_{\mathbf{i}} \exp \left(-\sum_{j=1}^{NREG} \mu_{ij} t_j\right)$$

$$D_{\mathbf{i}}(t_1, t_2, \dots, t_j + \Delta t_j, \dots, t_{NREG}) = C_{\mathbf{i}} \exp \left(-\sum_{j=1}^{NREG} \mu_{ij} t_j\right) \exp(-\mu_{ij} \Delta t_j)$$

Solving the above pair of equations for μ_{ii} results in

$$\mu_{ij} = \frac{1}{\Delta t_{j}} \ln \left[\frac{D_{i}(t_{1}, t_{2}, \dots, t_{j}, \dots t_{NREG})}{D_{i}(t_{1}, t_{2}, \dots, t_{j} + \Delta t_{j}, \dots t_{NREG})} \right]$$
(3)

The coefficients μ_{ij} are determined for all i and j in this manner.

The basic configuration data, that is, the set of thicknesses t_j , dose components D_i , and "attenuation coefficients" μ_{ij} constitute the required input data. For the set of base configuration, then, the coefficient C_i is calculated by the code OPEX-II from equation (2); that is,

$$C_i = D_i \exp\left(\sum_{j=1}^{NREG} \mu_{ij} t_j\right)$$

WEIGHT-MINIMIZATION PROCEDURE

The procedure for obtaining the minimum weight configuration by the steepest-descent method is presented in this section. The equations are basically from references 3 and 4. The narrative is expanded to illustrate the flow of computation and to comment on code output.

The mathematical problem to be solved is that of minimizing the weight w, a function of thicknesses t_j , while constraining the total dose D to some particular value; that is,

Minimize
$$w(t_1, t_2, \ldots, t_{NREG})$$

with constraints (a)
$$D = \sum_{i} D_{i} = \sum_{i} C_{i} \exp \left(-\sum_{i} \mu_{ij} t_{j}\right) = constant$$

(b)
$$t_i \ge 0$$

(c) t_{g} = constant for any desired values of ℓ

Constraint (b) ensures a physical solution. The optional constraint (c) is useful if it is desired that some thicknesses be kept from changing during the computation (e.g., the reactor core and reflector thicknesses). Constraint (c) is necessary for the spherical geometry programmed presently to prevent the trivial case of reducing reactor core size to zero.

An n-dimensional Euclidian vector space with Cartesian coordinates $t_1, t_2, \ldots t_{NREG}$ is defined. The following vectors are defined on this space:

$$\overline{\mathbf{t}} \equiv (\mathbf{t}_1, \ \mathbf{t}_2, \ \dots \ \mathbf{t}_{NREG})$$

$$\overline{\mathbf{g}} \equiv \left(\frac{\partial \mathbf{w}}{\partial \mathbf{t}_1}, \frac{\partial \mathbf{w}}{\partial \mathbf{t}_2}, \ \dots, \frac{\partial \mathbf{w}}{\partial \mathbf{t}_{NREG}}\right)$$

$$\overline{\mathbf{a}} \equiv \left(\frac{\partial \mathbf{D}}{\partial \mathbf{t}_1}, \frac{\partial \mathbf{D}}{\partial \mathbf{t}_2}, \ \dots, \frac{\partial \mathbf{D}}{\partial \mathbf{t}_{NREG}}\right)$$

The notation $\overline{t} = (t_1, t_2, \dots t_{NREG})$ means $t_1 \hat{X}_1 + t_2 \hat{X}_2 + \dots$ where \hat{X}_i are unit vectors in the i^{th} direction. Vectors \overline{g} and \overline{a} represent the gradient of weight and dose. The components of \overline{g} are evaluated from analytic expressions of weight as a function of thickness, and depend on geometry. The components of \overline{a} are evaluated from the partial derivatives of (1) and (2), namely,

$$\begin{split} \frac{\partial \mathbf{D}}{\partial \mathbf{t}_{k}} &= \sum_{\mathbf{i}=1}^{\mathbf{IMAX}} \frac{\partial \mathbf{D}_{\mathbf{i}}}{\partial \mathbf{t}_{k}} = \sum_{\mathbf{i}=1}^{\mathbf{IMAX}} \frac{\partial}{\partial \mathbf{t}_{k}} \left[\mathbf{C}_{\mathbf{i}} \exp \left(-\sum_{\mathbf{j}=1}^{\mathbf{NREG}} \mu_{\mathbf{i}\mathbf{j}} \mathbf{t}_{\mathbf{j}} \right) \right] \\ &= \sum_{\mathbf{i}=1}^{\mathbf{IMAX}} (-\mu_{\mathbf{i}k}) \mathbf{C}_{\mathbf{i}} \exp \left(-\sum_{\mathbf{j}=1}^{\mathbf{NREG}} \mu_{\mathbf{i}\mathbf{j}} \mathbf{t}_{\mathbf{j}} \right) \\ &= \sum_{\mathbf{i}=1}^{\mathbf{IMAX}} \mu_{\mathbf{i}k} \mathbf{D}_{\mathbf{i}} \end{split}$$

The unit vector u (see ref. 3 for derivation)

$$\hat{\mathbf{u}} = \frac{-\bar{\mathbf{g}} - \left(\frac{\bar{\mathbf{a}} \cdot \bar{\mathbf{g}}}{\bar{\mathbf{a}} \cdot \bar{\mathbf{a}}}\right)\bar{\mathbf{a}}}{\left[\bar{\mathbf{g}} \cdot \bar{\mathbf{a}} - \frac{(\bar{\mathbf{a}} \cdot \bar{\mathbf{g}})^2}{\bar{\mathbf{a}} \cdot \bar{\mathbf{a}}}\right]^{1/2}}$$

points in the direction of greatest weight decrease (steepest descent) along a hyperplane tangent to the hypersurface described by the equation

$$D(\overline{X}) = D(t_1, t_2, \dots, t_{NREG}) = Constant$$

Components of $\hat{\mathbf{u}}$, namely $\mathbf{u}_{\hat{\mathbf{j}}}$, represent increments of thickness to be added to each t_i to approach the minimum weight criterion.

The optimization code proceeds as follows:

- (1) A fraction f of each component u_j of \hat{u} is added to each thickness t_j . The fraction f is an input parameter. (A value of f = 1.0 has given satisfactory results.)
- (2) The new set of thicknesses generally does not return the correct dose constraint so a first-order correction is applied to each $\,t_{i}\,$ to return the dose constraint. This correction is

$$\overline{t}_{\text{new}} = \overline{t}_{\text{old}} + \left[D(\text{constraint}) - D(\text{calc}) \right] \times \frac{\overline{a}}{(\overline{a} \cdot \overline{a})}$$

Steps 1 and 2 are repeated until the relative change in weight from one iteration to the next is less than some prescribed value.

The code output includes final thicknesses and individual dose components as calculated from the dose-thickness relation for each iteration. It is incumbent on the user to make a final detailed proof calculation to verify the results of the prediction of the optimization code. Experiences have indicated that if input coefficients are determined accurately, final predictions of the optimization code are quite good provided the configuration is not radically changed If the configuration is changed severely, a recalculation of coefficients is in order.

SOME OTHER USES OF OPTIMIZATION CODE

Once an optimized base case is obtained, effects of nominal changes in reactor size, power level, and dose constraint on shield weight may be estimated using the optimization code. This is done by altering reactor radius, by scaling dose components proportionately, or by specifying a different dose constraint, respectively, and allowing OPEX-II to seek a new minimum weight configuration.

Cost optimization (minimization) may also be performed by specifying cost per unit volume rather than density (weight per unit volume) for each region.

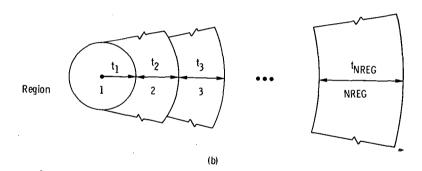
Input information to the code requires knowledge of each component of the dose. Output information also includes contribution of each component to the total dose. Thus one learns which regions are important and most sensitive to the calculation.

OPEX-II CODE

In this section the code details are presented, including

- (1) Flow chart for data input
- (2) FORTRAN IV listing
- (3) Sample problem and sample problem output

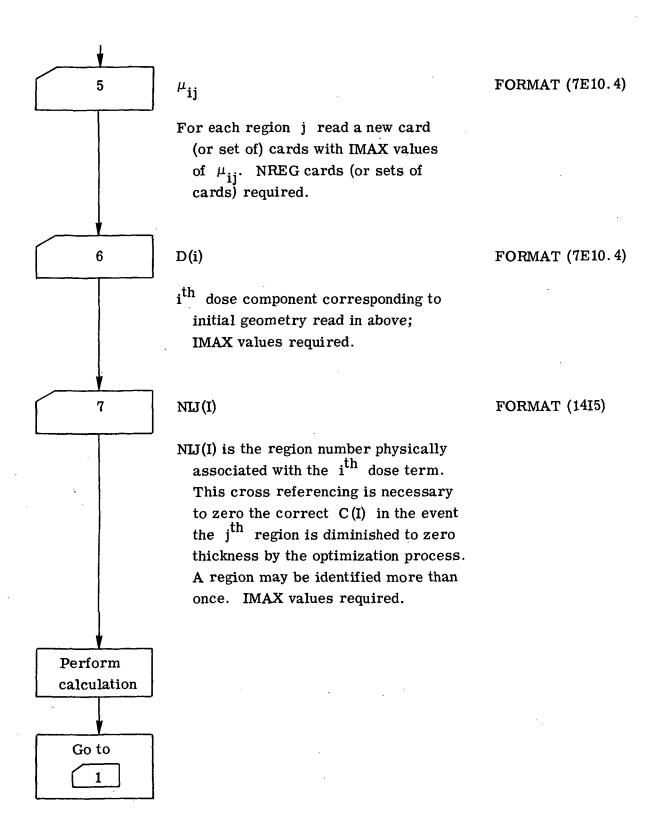
The geometry presently coded is spherical and is illustrated in sketch (b)



The data input consists of a set of thicknesses for each region. Radii are calculated internally. A thickness must be specified for each region.

Flow Chart for Data Input

	Contr	ol card		en grande en		
	Card column	Variable	FORMAT (315, 5F10.4)			
	1-5	NREG	Number of regions (thicknesses) in this problem (≤ 25)			
	6-10	MAX	Maximum number	r of iterations allowed		
	11-15	IMAX	Number of items	in dose equation (≤25)		
	16-25	DES	Desired dose rate	e (used only if > 0.0)		
	26-35	EPS	Convergence crite	erion for weight		
	36-45 EPSD		_	erion for initial dose (EPS, EPSD		
	46-55 CØN		Fractional step si	ize on $\hat{\hat{\mathbf{u}}}$ (0.5 \le CON \le 1.0)		
	56-65	CA	Fractional step si ≤ 1.0)	ize for initial dose $(0.5 \le CA)$		
1	,		•			
2) Thickness o	f j th region; equired.	FORMAT (7E10.4)		
		•				
	<u> </u>	•		•		
		Ø(J) Density o NREG values r	of j th region; required.	FORMAT (7E10.4)		
	t t	NB(J) = 0 cons o constant thic NB(J) = 1 allow	constraint flag train j th region ckness v j th region to values required.	FORMAT (2511)		



FORTRAN IV Listing

Presented in this section is the IBM-7094-II FORTRAN IV listing for the OPEX-II code. SUBROUTINE WEIGHT is coded for spherical geometry. Indication of how to alter this subroutine for the geometry of slabs bounded by a conical surface is given in that subroutine. The code can handle up to 25 regions and dose components as presently dimensioned. Running times are typically less than 0.1 minute on the IBM 7094-II for 12-region, 16-dose component problems.

Sample Problem

A sample problem consisting of a reactor, molybdenum reflector, and a shield consisting of seven alternating layers of lithium hydride and tungsten is illustrated in figure 1. Region descriptions, densities, and initial thicknesses (guessed) are listed in table I. The configuration is to satisfy a dose constraint of 2 mrem per hour at a point 20 meters distant. A series of discrete ordinates calculations for both neutrons and gammas was made to calculate doses from each source; perturbations were made to determine the attenuation coefficients $\mu_{\bf ij}$. The results of these calculations are listed in tables II and III. Because the core radius, plenum, pressure vessel, and reflector thicknesses are to be constrained in this calculation, a value of $\mu_{\bf ij}$ = 0.0 is arbitrarily assigned to these regions.

The complete computer output for this OPEX-II calculation to seek a minimum weight for this seven-layer configuration is given in table IV. The output consists, first, of a listing of all input information, followed by the value of the dose constraint (DES), the value of the calculated dose for the initial configuration (DOS), and the weight (WT) in grams of the initial configuration. If DES and DOS do not agree to within the parameter EPSD, a new set of thicknesses is calculated and printed; this new set of thicknesses satisfies the dose constraint. The listing of DES, DOS, and WT is followed by the values of the dose components D(I) and the thickness of each region T(I).

The results of each OPEX-II iteration for the present problem are shown in figure 2. Shown, for each iteration, is the size and relative position of each of the layers as adjusted in the course of the calculation. The final thicknesses and values of the dose components are listed in tables I and II for comparison with initial values. The initial, guessed configuration weighed 3.594×10⁷ grams (79 000 lb); the final, 2.999×10⁷ grams (66 000 lb).

TABLE I. - REGION DESCRIPTION .

Region j	Description	Density, g/cm ³	Thickness, cm (initial guess)	Final prediction of shield thicknesses, cm
i	Reactor core	9.957	26.0 radius	
2	Plenum	8.647	2.50	
3	Pressure vessel	16.763	. 60	
4	Molybdenum reflector	9.234	11.00	
5	Lithium hydride	. 75	(17. 90)	20. 52
6	Tungsten	19.3	(7.00)	9. 71
7	Lithium hydride	. 75	(14.00)	12. 32
8	Tungsten	19.3	(5.00)	2.82
9	Lithium hydride	. 75	(10.00)	10. 32
10	Tungsten	19.3	(3.50)	2.33
11	Lithium hydride	. 75	(59.50)	39.29

TABLE II. - DOSE COMPONENTS

Region j	Dose component, Value of dose compone D_i , D_i , $mrem/hr$				
		With initial shield thicknesses	With final shield thicknesses		
1	Neutron	0.02430	0.2527		
2	Core gamma	. 00303	. 0079		
3	Plenum, pressure vessel capture gamma	.00196	. 0049		
4	Plenum, pressure vessel inelastic gamma	. 00220	. 0055		
5	Reflector capture gamma	. 204	. 4782		
6	Reflector inelastic gamma	. 00504	.0129		
7	Region 6 tungsten capture gamma	. 0921	. 4007		
8	Region 6 tungsten inelastic gamma	. 00974	. 0949		
9 .	Region 8 tungsten capture gamma	. 0988	. 2696		
10	Region 8 tungsten inelastic gamma	.0278 .0759			
11	Region 10 tungsten capture gamma	. 201	. 2650		
12	Region 10 tungsten inelastic gamma	.0947 .1320			
	Total	0. 7647	2.000		

TABLE III. - COEFFICIENTS

Core Plenum n, γ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	4 5 Plenum Reflector n, n'γ n, γ Coe 0 0 0 0 0 0	6 7 tor Reflector W(6) n, n' γ n, γ Coefficient, μ_{ij} , cm ⁻¹ 0 0 0 0 0 0 0 0 0 0 0 0 0 0	W(6) W(6) n, \(\gamma \) cm ⁻¹ 0 0	8 W(6) n, n'Y	9 W(8) n, γ	10 W(8) n, n'Y	11 W(10) n, γ	12 W(10) n, n' \(\cdot \)
Core Plenum n, γ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	n, 7	or Reflector n, n'\gamma coefficient, \(\mu_{1j}\) 0 0 0	W(6) n, γ l, cm ⁻¹ 0 0	W(6) n, n'γ	w(8)	W(8)	W(10) n, γ	W(10)
9amma n, γ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	n, 1	0 0 0	n, γ cm ⁻¹	n, n'y	n, γ	n, n'y	n, γ	n, n' y
0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 1347 .0189 .0147 .2480 .774 .772 .1369 .0228 .0204 .2316 .770 .766		oefficient, μ_{ij}), cm ⁻¹					
0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 1347 .0189 .0147 .2480 .774 .772 .1369 .0228 .0204 .2316 .770 .766	0 0 0	0 0 0	0 0 0					
0 0 0 0 0 0 0 0 0 0 0 0 0 3 0 0 0 1347 .0189 .0147 .2480 .774 .772 .1369 .0228 .0204 .2316 .770 .766	0 0	0 0	0 0	0	0	0	0	0
0 0 0 0 0 0 0 0 0 0 0 0 1347 .0189 .0147 .2480 .774 .772 .1369 .0228 .0204 .2316 .770 .766	0	0		0	0	0	0	0
.2480 .774 .772 .1369 .0228 .0204 .2316 .770 .766			_	0	0	0	0	0
. 0189	0	0	0	0	0	0	0	0
. 774	.0147 .0212	. 024	. 3629	. 2297	. 2018	. 1885	.1837	. 1759
. 0228	. 772 . 805	. 795	. 2719	. 1812	. 2386	. 2536	. 2543	. 2577
. 770	.0204 .0167		.0210	.0250	. 2257	. 1983	. 1910	.1795
760	. 766	. 781	. 798	. 827	. 1779	. 1255	. 2052	.2368
_	.022 .0201	•	.0187	.0245	. 0230	.0250	. 2231	. 1808
.2306 .768 .763	. 763 . 805	. 776	. 782	. 812	. 810	. 851	.2407	.1107
.1187 .0244 .0226	.0226 .0196	. 0252	.0232	.0286	. 0232	.0291	. 0231	.0297

TABLE IV. - OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

```
$18FIC OPT
           MAIN PROGRAM---ADOPTED FROM OPEX (AI) DIST. BY RSIC/ORNL
           GPFX BASIC EQUATIONS COMPLETELY REPROGRAMMED BY-
                                  G.P.LAHTI **NASA-LEWIS RESEARCH CENTER
C
                                                          CLEVELAND, OHIG
C
                                NREG, MAX. IMAX. DES. CUN, CA, EPS, EPSD, WI. GG. AG. II. NB(25). NIJ(25), A(25). C(25). EMU(25.25).
           COMMON
         I COS. AA.
         2 [[25].
          3 11251.
                                 RHO125).
                                                       G125).
  7777 CALL INPUT
          CALL WEIGHT
           DES -67. 0.0') CALL INIT
DS = 00S
D0 120 IT =1, MAX
           W = WT
           SU = SORTIGG -AG*AG/AA)
AGA= AG/AA
           DO 60 I =1. NREG
           If ( NR(I) .EO. U) GU TO 60

U(I) =(-C(I) + AGA * A(I) )/ SU

T(I) = T(I) + CON * U(I)

If( T(I) .EE. 0.0) CALL CLEAR(I)
     60 CONTINUE
           CALL DOSE
CONST = (DS-DUS)/AA
DC EC I=1.NREG
           IF( NH(1) .EO. 0) GU TO BG
T(1) = 1(1) + CGNST * A(1)
IF( T(1).LE. U.O) CALL CLEAR(1)
     80 CONTINUE
           CALL WEIGHT
           WRITE (6.60C) 11. W1. DGS. (D(1). I=1.IMAX)
WRITE (6.601) (T(1). I=1.NREG)
IF( ABS(NT-W)/ NT -EPS) 110. 110. 120
    120 CONTINUE
   11C GG TO 7777
600 FERMATCHEC/7HO IT = 13.7X.5HWF = 1PE12.4.7X.6HCOS = 1PE12.4/
              7HGD(1)= 1P9E12.4/(7X.1P9E12.4))
   601 FORMAT(7FOT(I)= 1PSE12.4/(7x.1PSE12.4))
           END
SIBFIC INPUTT DECK
           STEROUTINE INPUT
                             NREG. MAX. IMAX. DES. CGN, CA. EPS. EPSD. *
WI. GG, AG. II. NB(25). NIJ(25).
A(25). C(25). EMU(25.25).
           CEMMON
         1 COS. AA.
         2 1125). A125). C(25). U(25). U(25).
READ(5,4UC) NREG. MAX. IMAX. DES. EPS. EPSD. CON. CA
WRITE(6,500) NREG. MAX. IMAX. DES. EPS. EPSD. CON. CA
          REAC (5.401) (T(1), I=1.NREG)

REAC (5.401) (RHU(1), I=1.NREG)

REAC (5.403) (NB(1), I=1.NREG)

WRITE(6.510) ( 1, I(1), RHC(1).NB(1), I = 1. NREG)
           WRITE(6.550)
       WRITE(2-550)

CC 7 J=1.NREG

RFAC (5-401) (EMU(1-J). I=1.IMAX)

7 WRITE (6-57() J.(EMU(1-J). I=1.IMAX)

RFAC (5-401) (D(1). I=1.IMAX)

RFAC (5-402) (NIJ(1). I=1.IMAX)

NIJ(1) IS THE REGICN NUMBER ASSOCIATED WITH THE ITH DOSE TERM

CC 9 I=1.IMAX

RE = 0.0

CC 8 I=1.NMEG
           DC 8 4=1.0KEG
       8 BF = 68 + EMU(I,J)*I(J)

9 C(I) = D(I)*EXP(88)
          WRITE(6,530) (I. C(1), D(1), NIJ(1), I=1, IMAX)
   400 FCFMAT(315. 5F10.4)
401 FCFMAT(7F1C.4)
    402 FERMAT(1415)
    403 FCRMAT(2511)
   4U3 FERMAI(251)

500 FERMAI(9F1 AREG = 13/ SH MAX = 13/ 9H IMAX = 13/ 9H DES =

1 1PF12.4/ SH EPS = 1PE12.4/ 9H EPSD = 1PE12.4/ 9H CON =

2 1PE12.4/ SH CA = 1PE12.4/

510 FERMAI(1H0/34HCKEGIGN I(1) RHC(1) NB(1)/(17.2F10.3.17

530 FERMAI(1HC/34HC 1 C(1) D(1) AIJ/(15.1P2E12.4.

550 FERMAI(13ChCREGICN-J MU(1.J) )
                                                                          RHC(I) NB(I)/(17,2F10.3,17))
                                                                  D(I) NIJ/(15-1 P2E12-4-15))
MU(I-J)
    57G FCRMAT(16. 3x. 1P9E12.4/(5x.1P9E12.4))
           RETURN
```

ENC

TABLE IV. - Continued. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

SIBFIC INTIXX CECK SLAROULINE INIT NREG, MAX. IMAX. DES. CON. CA. EPS. EPSD. NT. GG. AG. 11. NB(25). NIJ(25). A(25). C(25). EMU(25.25). CEMMON NREG. BG. NT. GG. A(25). AG. 11. C(25). I COS. AA. 2 E(25). G(25). G(25). WRITE(6,170) DES. DOS. NT. (D(1). I=1,1MAX) Q= 1.0 CC 95 K =1.MAX CCSA = GOS A/A = SORT(AA) CINSI = (A*W /AAA IF (CDS-DES) 62. 10. 60 60 CCNST = -CGNST 62 CC 70 I=1.NREG IF(NH(I) .EU. O) GC TO 7C T(I) = T(I) + CGNST * A(I) IF(T(I).LE. U.O) CALL CLEAR(I) 7C CENTINUE CALL DUSE IF(AMS(COS-DES)/DES .LE. EPSD) GO TO 10 IF(((005-DES)*(DOSA-CES)) .LE. U.O) Q = 0.5 * Q S5 CENTINUE K = 777 10 CALL WEIGHT CALL DOSE WRITE(6.1CC) K WRITE(6-100) K 100 FCRMAT(22F0INITIAL IHICKNESSES FCUND AFTER 13-11H ITERATIONS) WRITE(6-160) (T(1), I=1.NREG) 160 FCRMAT(21F0CALCULATED INITIAL THICKNESSES/ (1P9E12-4)) WRITE(6-170) DES. DOS. WT. (D(1), I=1.IMAX) 170 FCRMAT(7FCCES = 1PE12-4-6X, 6FDCS = 1PE12-4-6X, 5HWT = 1PE12-4/ 1 7F0D(1) = 1PSE12-4/ (7X-1PSE12-4)) RFTURN END END

SIBFTC CLEANX BECK

```
Steroutine Clear(J)
CCMMON NREG. MAX, IMAX, DES. CCN. CA. EPS. EPSD.
1 EDS. AA. WI. GG. AG. 11. NB(25). NIJ(25).
2 E(25). A(25). C(25). EMU(25.25).
3 T(25). RHG(25). G(25). U(25)
C THE JTH REGION HAS JUST BEEN WIPED OUT
T(J) = O.C
NE(J) = C
DC 5 [=1.1MAX
1F( J.EO. NIJ(I) ) C(I)= C.O
5 CINTINUE
RETURN
ENC
```

TABLE IV. - Continued. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

\$16FTC FOSEXX CECK

```
SIEROUTINE CUSE

CCMMUN NREG. MAX. IMAX. DES. CON. CA. EPS. EPSD.

1 COS. AA. WT. GG. AG. 11. NB(25). N1J(25).

2 C(25). A(25). C(25). EMU(25.25).

3 1(25). RHO(25). G(25). U(25)

CCS = C.C

DF 40 1=1.IMAX

BF = 0.C

CC 30 J=1.NKEG

30 Bf = BB + EMU(1.J)* T(J)

D(1) = C(1) * EXP(-BB)

40 DCS = LOS + D(1)

CCS = LOS + D(1)

CAJ = A.A

AC = A.G

AJ = A.A

AC = A.G

AJ = O.O

DC 60 K = 1.NKEG

A(K) = 0.0

DC 60 K = 1.NKEG

A(K) = 0.0

DC 60 K = 1.IMAX

50 A(K) = A(K) = EMU(1.K) * D(1)

AJ = AA + A(K)**2

AG = AG + A(K)*G(K)

60 CCNTINUE

RETURN
```

SIBFIC MISPHR DECK

```
SUBROUTINE WEIGHT
CCOING FOR UPEX SHIELDING OPTIMIZATION CCDE
SPERICAL SHELL GEOMETRY — G.P.LAHTI — NASA-LEWIS
FCR PLANE SLABS BUUNDED BY CONE OF HALF ANGLE THETA,
THIS SUBROUTINE MAY BE USED SIMPLY BY REPLACING
THE TERM FORPI BY 4*(IAN(THETA))**2
             CATA FURP'I / 12.56636/
DIMENSION ROUT(25)
           CCMMON
1 EOS. AA.
2 E(25).
                                    NREG. MAX. IMAX. DES. CON. CA. EPS. EPSD. WT. GG. AG. II. NB(25). NIJ(25). A(25). C(25). EMU(25,25).
                                       RHO (25).
                                                                 6(25).
           3 1(25).
                                                                                            U(25)
             CALCULATE OUTER RADIUS OF EACH REGION, AND WEIGHT
c
            R(U1(1)* T(1)
RRR= RULT(1)**3
WEIGHT=- RHO(1)*RRR
CC 5 .I=2.NREG
            K2= RRH
RCUT(I)= ROUT(I-1)+T(I)
KFK= ROUT(I)**3
        RMR= ROUTITIONS

WEIGHTSCHO(L)*(RRR-R3)

NI= WEIGHTSCHOI/3.

CALCULATE PARTIAL NI DERIVATIVES (DW/DXI) = G(I).

G( = G.G
GC=C.J
CC 9 I=1.NREG
            RR= C.Q
             G(1) = 0.C
             IF( NH( I) .EQ. () GG TG 5
CL & J=1.NREG
             R2= RR
RK= ROLT(J)**2
        8 G(1) = G(1) + RHO(J)*(KK-R2)
G(1) = G(1) * FGRP1
GC = GG + G(1)**2
         S CENTINUE
             RETURN
END
```

TABLE IV. - Continued. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

```
NREG = 11
 MAX =.
         50
 IMAX =
         12
 CES =
EPS =
          2.0000E 00
1.0000E-03
           1.0000E-03
 EPSC =
 CON =
           7-06C0E-01
 CΔ
           7.00C0E-01
REGION
             T(1)
                      REC(I) NB(I)
           26.000
                       9.957
            2.500
                       £.647
                                   C
     3
            0.660
                      16.763
                                   C
           11.000
                       9.234
                                   C
           17.900
                       €.756
            7.000
                      15.300
           14.000
                       C.750
                                   1
            5.000
                      15.300
           10.000
    10
           3.5CC
59.5CC
                      19.300
    11
                       C.750
REGION-J
                     KL.IJUM
                        0.
                                                                                                                     0.
    1
           0.
                                     0.
                                                   ٥.
                                                                ٥.
                                                                             0.
                                                                                           0.
                                                                                                        0.
           0.
                                     с.
    2
                                                   ٥.
                                                                o.
                                                                             ٥.
                                                                                           ٥.
                                                                                                        ٥.
                                                                                                                     ٥.
           0.
                        0.
    3
                                                                                                                     ٥.
           c.
                        0.
                                     С.
                                                                             0.
                                                                                                        0.
    4
           C.
                        C.
                                                   0.
                                                                0.
                                                                             0.
                                                                                           0.
                                                                                                        ο.
                                                                                                                     0.
           0.
                        0.
                                     C.
    5
           1.3470E-C1 1.89UCE-C2
                                     1.47COE-02
                                                                                          3.6290E-01 2.2970E-01
                                                   1.4700E-02
                                                                2.1200E-02 2.4000E-02
                                                                                                                     2.0180E-01
                        1.8370E-01
7.74C0E-01
                                     1.759CE-01
7.7200E-01
           1.8E50E-C1
    6
                                                   7.7200F-01 8.0500E-01 7.9500E-01 2.7190E-01 1.8120E-01 2.3860E-01
           2.48CCE-C1
           2.5360E-01
                        2.5430E-01
2.28C0E-02
                                     2.5770E-01
    7
           1.369GE-01
                                     2.0400E-02
                                                   2.0400E-02 1.6700E-02 2.0600E-02 2.1000E-02 2.5000E-02
                                                                                                                     2.2570E-01
           1.9830E-01
                        1.9100E-01
                                     1.795 UE-01
           2.3160E-C1
                        7.70CCE-01
                                                   7.6600E-01 7.9000E-01 7.8100E-01 7.9800E-01 8.2700E-01 1.7790E-01
    R
                                     7-660CE-01
           1.255CE-01
                        2.0520E-C1
                                     2.3680E-01
    ç
           1.2180E-C1
                        2.40C0E-02
                                     2.2000E-02
                                                   2.2000E-02 2.0100E-02 2.3900E-02 1.8700E-02 2.4500E-02 2.3000E-02
           2.50CCE-C2
                        2.2310E-01
                                     1.8080E-01
   10
           2-3060F-01
                        7.6800F-C1
                                     7.6300E-01
                                                   7-6300F-01 8-0500F-01 7-7600F-01 7-8200F-01 8-1200F-01 8-1000F-01
           8.51COE-01
                        2.4070E-01
                                     1.1670E-01
           1.1870E-01 2.440CE-02 2.2600E-02
2.91CCE-02 2.31CCE-02 2.570GE-02
   11
                                     2.2600E-02
                                                   2.2600E-02 1.9600E-02 2.5200E-02 2.3200E-02 2.8600E-02 2.3200E-02
             C(1)
                          DII) NIJ
      2.9425E G5
4.9449E G3
                   2.43COE-C2
                    2.0300E-C3
      2.3557E C3
                   1.9600E-C3
2.2000E-03
      2.6536E (3
      3.5981E 05
                    2.04COE-01
      1.1522E (4
                    5.04C0E-03
   6
      2.1973E (6
2.2521E 04
                    9.21CUE-C2
                                   6
                    5.7400E-03
      5.5079E C4
2.0543E C4
                    5.8800E-02
2.78CUE-02
  10
      1.1039E 05
                    2.01C0E-C1
      2.8426E C4
                    5.47COE-C2
CES =
        2.00CCE 0C
                          DOS =
                                   7.6467E-01
                                                             3.5939E 07
        2.4300E-02 3.0300E-03 1.9600E-03 2.2000E-03 2.0400E-01 5.0400E-03 9.2100E-02 9.7400E-03 9.8800E-02 2.7800E-02 2.0100E-01 5.47CCE-02
E(1)=
INITIAL THICKNESSES FOUND AFTER 11 LIERATIONS
CALCULATED INTITIAL THICKNESSES
 2.60CCF C1 2.500CE C0 6.CCCCE-01 1.1000E 01 1.77C1E 01 6.5022E 00 1.3850E 01 4.4536E 00 5.8800E 00 2.8226E CC 5.9468E 01
        2.COOGE CG
                          DCS = 2.00(3E.00
                                                            3.2192E 07
        3.8560E-C2 1.1539E-C2 7.4045E-03 8.3112E-03 8.1703E-01 1.9633E-02 2.9957E-01 3.0622E-02 2.2937E-01
        6.4559E-C2 3.2958E-C1 1.4374E-C1
```

TABLE IV. - Continued. OPEX-II LISTING AND OUT PUT FOR SAMPLE PROBLEM

```
₩7 = 3.1542E C7
 17 =
                                          COS = 2.00C5E 00
         4.1951E-C2 9.6445E-03 6.17(9E-03 6.9265E-03 6.7567E-01 1.6311E-02 3.4357E-01 3.5056E-02 2.6938E-01 7.48(7E-C2 3.6601E-C1 1.5453E-01
C(1)=
         2.60CGE C1 2.50CUE CO 6.COCUE-U1 1.10CUE C1 1.736CE C1 6.8982E CO 1.3516E C1 4.4422E CO 9.6920E CO 2.6959E CC 5.9234E C1
T(I)=
 11 =
                 FT = 3.1772E C7
                                        DOS = 2.0064E 00
        4.5363E-02 7.5752E-03 5.0914E-03 5.7148E-03 5.5197E-01 1.3414E-02 3.8134E-01 3.8937E-02 2.9864E-01 8.1425E-02 4.0285E-01 1.6771E-01
C(I)=
        2.6000E C1 2.5000E C0 6.001CE-C1 1.1000E C1 1.7056E 01 7.2748E 00 1.3174E 01 4.3829E 00 9.5066E 00 2.6602E CC 5.8920E 01
3 ( 1 ) =
                 ₩T = 3.1652E C7
                                          DOS = 2.0004E 00
 IT =
        4.5506E-02 6.854IE-03 4.363IE-03 4.8974E-03 4.6772E-01 1.147IE-02 4.0820E-01 4.190IE-02 3.0352E-01
C(1)=
        H.C718E-02 4.3785E-01 1.8335E-01
        2.6000E C1 2.5000E C0 6.0000E-01 1.1000E C1 1.6848E 01 7.5684E 00 1.2842E 01 4.23(9E 00 9.3440E 00 2.7466E 00 5.6461E 01
I(1)=
 IT =
                 ¥T = 3.1553€ €7
                                          DUS = 2.0003E 00
        5.4284E-C2 6.4299E-03 4.0862E-03 4.5665E-03 4.3372E-01 1.0738E-02 4.1939E-01 4.3600E-02 2.8580E-01 7.4048E-02 4.6407E-01 1.5958E-01
C(1)=
        2.6000E C1 2.5000E C0 6.0000E-01 1.1000E 01 1.6803E 01 7.7148E 00 1.2566E 01 4.01C9E 00 9.2477E 00 2.5344E 0C 5.7935E C1
 IT =
                 FT = 3.1463E C7
                                          DOS = 2.00G3E 00
        5.9147E-02 6.3109E-03 4.0056E-C3 4.4961E-03 4.2181E-01 1.0526E-02 4.1955E-01 4.4555E-02 2.6939E-01 6.8444E-02 4.60C4E-C1 2.12C5E-C1
0(1)=
        2.6000E C1 2.500CE C0 6.0000E-01 1.1000E C1 1.6870E 01 7.7951E 00 1.2329E 01 3.8219E 00 9.2176E 00 3.0927E 00 5.7334E C1
T(1)=
 IT =
                 w1 = 3.1379E 07
                                          DOS = 2.0003E 00
        6.3855E-02 6.2866E-03 3.9862E-03 4.4743E-03 4.1706E-01 1.0472E-02 4.1628E-01 4.5488E-02 2.6057E-01 6.5357E-02 4.6658E-01 2.1951E-01
E(11) =
        2.60000 01 2.50000 00 6.00000-01 1.10000 01 1.6994E 01 7.8688E 00 1.2120E 01 3.6802E 00 9.2390E 00
T(1)=
        3.1880F 00 5.6704F 01
                                         DOS = 2.0003E 00
 IT =
                 kT = 3.12996 07
        6.8410E-C2 6.3186E-C3 4.0029E-03 4.4931E-03 4.1657E-01 1.0510E-02 4.1273E-01 4.6603E-02 2.5675E-01 6.3876E-02 4.6705E-01 2.2256E-01
G(1) =
        2.6000E 01 2.500CE CO 6.0000E-01 1.1000E 01 1.7147E 01 7.9435E 00 1.1940E 01 3.5656E 00 9.2955E 00 3.2372E 00 5.6063E 01
T(1)=
 11 =
                WT = 3.1221E C7
                                          EOS = 2.00C3E 00
        7.2956E-C2 6.3764E-G3 4.0363E-O3 4.5306E-O3 4.1809E-O1 1.0591E-O2 4.0941E-O1 4.7846E-O2 2.5581E-O1 6.3317E-O2 4.6337E-C1 2.2358E-C1
D(1)=
        2.60C0E C1 2.50CCE (C 6.CUOUE-O1 1.1000E 01 1.7311E 01 8.0188E 00 1.1784E 01 3.4798E 00 9.3693E 00 3.2583E 0C 5.5416E C1
T(1)=
                WT = 3.1146E C7
                                         COS = 2.0003E 00
11 =
        C(1)=
        2.80CCE 01 2.50CCE CC 6.COOUE-01 1.10QOE 01 1.7478E 01 8.0930E 00 1.1646E 01 3.4035E 00 9.4497E 00 3.2641E CC 5.4764E 01
```

: :

TABLE IV. - Continued. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

		*							
IT =	10 ¥ T	= -3.1073E	C7 DCS	= 2.0003E	00		•	•	
G(1)=		6.5201E-03 4.7113E-01	4.12C9E-03 2.2261E-01	4.6255E-03	4.2335E-01	1.0801E-02	4.0332E-01	5.0475E-02	2.5747E-01
T(1)=		2.50CCE CC 5.41C7E 01	6.000UE-U1	1.1000E 01	1.7641E 01	8.1653E 00	1-1525E 01	3.3365E OC	9.5312E 00
IT =	11 +1	= 3.1002E	c7 968	= 2.0GC3E	co				
C(I)=		6.5948E-03 4.6402E-01	4.1650E-C3 2.2111E-01	4.675UE-03	4.2625E-01	1.09116-02	4.0058E-01	5.1818E-02	2.5889E-01
T(1)=		2.5000E 00 5.3444E Cl	6.00COE-01	1.1000E 01	1.7801E 01	8.2355E 00	1.1419E 01	3.2765E 00	9.6112E 00
17 =	12 11	= 3.C933E	c7 pes	= 2.00(3E	00				
=(1)3		6.6086E-03 4.5660E-01	4.2084E-03 2.1929E-01	4.7238E-03	4.2911E-01	1.1020E-02	3.9809E-01	5.3174E-02	2.6038E-01
=(1)F		2.500CE 00 5.2778E C1	6.CUCCE-C1	1.1000E 01	1.7954E 01	8.3039E 00	1.1327E 01	3.2220E 00	9.6886E 00
17 =	13 11	= 3.08678	07 DOS	= 2.0003E					
E(I)=		6.7407E-C3 4.4E55E-C1	4.2507E-03 2.1721E-01	4.7712E-03	4.3189E-01	1.1126E-02	3.9585E-01	5.4544E-02	2.6184E-01
=(1)T		2.5000É 00 5.2107E 01	6.CCC0E-01	1.1000E 01	1.8102E 01	8.3704E 00	1.1250E 01	3.1726E 00	9.7628E 00
IT ≠		= 3.CEC2E		= 2.0003E					
C(1)=			4.2917E-03 2.1487E-01	4.8172t-03	4.3456E-01	1 • 12 30E-02	3.9386E-01	5.5931E-02	2.6324E-01
T(1)=		2.500GE 0C 5.1433E 01	6.C0C0E-01	1.1000E 01	1.8244E U1	8.4355E 00	1.1187E 01	3.1277E 00	9.8334E 00
1) =	15 h f	= 3.074CE	C7 DUS	= 2.00C3E	00				
C(1)=		6.8788E-03	4.3312E-03 2.1234E-01	4.8616E-03	4.3712E-01	1.1331E-02	3.9214E-01	5.7340E-02	2.6457E-01
1(1)=		2.5000E 00 5.6756E C1	6.COLUE-01	1.10006 01	1.8381E 01	8.4992E 00	1.1137E 01	3.0871E CO	9.9003E .00
	14	- 2.64.505		- 2 00000	00				
IT = [(1)=		= 3.068CE 6.5448E-03	4.3655E-U3	= 2.0063E 4.9046E-63		1.1429F-02	3.9068E-01	5.8777E-02	2.6580E-01
	6.5823E-02	4.2487E-01	2.0954E-C1						
1(1)=		2.5000E CC 5.0077E C1	6.60008-01	1.1000E 01	1.85146 01	8.5620E 00	1.1102E 01	3.0504E 00	9.9634E 00
IT =	17 ≱T	= 1 3.C622E	07 005	= 2.0003E	00				
£.(7.0050E-03 4.1647E-01	4.4066E-03 2.6652E-61	4.9401E-03	4.4193E-01	1.1524E-02	3.8947E-01	6.0250E-02	2.6694E-01
f(1)=		2.500CE CU 4.9396E 01	6.00008-01	1.1000E 01	1.8042E 01	8.6240E 00	1.1079E 01	3.0176E 00	1.0023E 01

TABLE IV. - Continued. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

```
IT = .18
                MT = 3-0566E 07
                                       DOS = 2-0003E 00
        1.3177E-01 7.0714E-03 4.4425E-03 4.5865E-03 4.4421E-01 1.1617E-02 3.8852E-01 6.1765E-02 2.6798E-01
        6-6849E-02 4-0787E-01 2-0324E-01
        2.6000E 01 2.5000E 00 6.0000E-01 1.1000E 01 1.8767E 01 8.6857E 00 1.1070E 01 2.9882E 00 1.0078E 01 3.1205E 00 4.6713E 01
 IT - 10
                MT = 3-0512E 07
                                         DOS = 2.0003F 00
        1.3931E-01 7.1323E-03 4.4775E-03 5.0257E-03 4.4642E-01 1.1707E-02 3.8783E-01 6.3332E-02 2.6891E-01 6.7392E-02 3.9906E-01 1.5572E-01
6(1)=
        2.60C0E 01 2.50C0E 00 6.00C0E-01 1.10C0E 01 1.8889E 01 8.7473E 00 1.1074E 01 2.9622E 00 1.0129E 01 3.05C4E (C 4.8030E 01
T(1)=
                hT = 3.046CE C7
                                       DOS = 2.0003E 00
       1.4714E-01 7.1919E-C3 4.5116E-03 5.0640E-03 4.4858E-01 1.1795E-02 3.8738E-01 6.4962E-02 2.6974E-01
0(1)=
        6-7954F-02 3-9004F-01 1-9596F-01
        2.60CCE C1 2.5000E C0 6.COCCE-01 1.1000E 01 1.9008E 01 8.8092E 00 1.1092E 01 2.9391E 00 1.0175E 01
1(1)=
        3-0571F OC 4-7346F OL
 TT = 21
                kT = 3.04CSE C7
                                         DGS = 2.0003E 00
        1.5526E-(1 7.2505E-03 4.5451E-C3 5.1016E-03 4.5071E-01 1.1882E-02 3.8718E-01 6.6664E-02 2.7045E-01 6.8534E-02 3.8CE1E-C1 1.9194E-01
C(1)=
        2.60CCE (1 2.50COE 00 6.CUCCE-01 1.10COE 01 1.9126E 01 8.8717E 00 1.1122E 01 2.9189E 00 1.0217E 01 3.0205E CC 4.6662E 01
1(1)=
 IT = 52
                WT = 3.0361E C7
                                         DOS ≈ 2.0003E 00,
        1.6363E-01 7.3083E-03 4.5781E-03 5.1386E-03 4.5282E-01 1.1968E-02 3.8723E-01 6.8452E-02 2.7105E-01 6.9131E-02 3.7136E-01 1.8766E-01
C(1)=
        2.60CCE (1 2.5000E 00 6.C0CCE-01 1.1000E 01 1.9243E 01 8.9352E 00 1.1165E 01 2.9012E 00 1.0255E 01 2.9012E 01 4.5579E 01
I(I)=
                                       COS = 2.0003E 00
11 = 23
               ₩1 = 3.0315E 07
       1.7223F-01 7.3656E-03 4.6108E-03 5.1753E-03 4.5493E-01 1.2052E-02 3.8752E-01 7.0340E-02 2.7152E-01 6.9744F-02 3.6168E-01 1.8314E-01
f(11)=
        2.60C0E 01 2.50CUE CO 6.COUNE-01 1.10COE 01 1.9360E 01 9.00COE 00 1.1222E 01 2.8859E 00 1.0267E 01
        2.9362E UC 4.5256E C1
                WT = 3.0271E C7
                                        COS = 2.0003E 00 .
        1.8104E-C1 7.4227E-03 4.6434E-03 5.2120E-03 4.5707E-01 1.2137E-02 3.8806E-01 7.2343E-02 2.7187E-01
        7.0373E-02 3.5179E-C1 1.7836E-01
       2.6060E (1 2.5060E 00 6.0000E-01 1.1000E 01 1.9477E 01 9.0666E 00 1.1291E 01 2.8726E 00 1.0314E 01
        2.8882E 00 4.4615E 01
                FT = 3.0229E C7
                                      COS = 2.0003E 00
 II = 25
      1.9002E-C1 7.4800E-03 4.6762E-03 5.2488E-03 4.5926E-01 1.2221E-02 3.8883E-01 7.4478E-02 2.7209E-01
C(1)=
        7.1017E-02 3.4166E-01 1.7333E-01
T(1)= 2.6000E C1 2.500CE CC 6.0000E-01 1.1000E C1 1.9595E 01 9.1353E 00 1.1373E 01 2.8612E 00 1.0336E 01 2.8360E CC 4.393EE C1
```

TABLE IV. - Concluded. OPEX-II LISTING AND OUTPUT FOR SAMPLE PROBLEM

MT = 3.01696 67 DOS = 2.0003E 00 IT - 26 1.5912E-C1 7.5379E-03 4.7054E-03 5.2860E-03 4.6153E-01 1.2306E-02 3.6985E-01 7.6764E-02 2.7217E-01 7.1675E-02 3.3131E-C1 1.6806E-01 = (130 2.6000E 01 2.50CGE CO 6.0000E-01 1.1000E 01 1.9715E 01 9.2064E 00 1.1469E 01 2.8515E 00 1.0353E 01 T(1)= 2.7791E GO 4.3259E GL nos = 2.0003F 00 1.1 = 3.01€CE C7 11 - 27 2.CH29E-C1 7.5966E-03 4.7432E-03 5.3240E-03 4.6389E-01 1.2391E-02 3.9110E-01 7.9222E-02 2.7211E-01 1.2341E-(2 3.2075E-C1 1.6255E-01 2.6ULOE 01 2.5UCOE CO 6.0UGUE-01 1.1000E C1 1.9838E 01 9.2805E 00 1.1577E 01 2.8433E 00 1.0363E 01 2.7175E 0C 4.2586E CI T(1)= ⊌T = 3.0114E C7 005 = 2.0063 F 00 IT = 28 7.1746E-(1 7.6568E-03 4.7780E-03 5.3631E-03 4.6638E-01 1.2479E-02 3.9259E-01 8.1875E-02 2.7190E-01 0(1)= 7.3031E-02 3.C958E-C1 1.5682E-C1 2.600CE 01 2.500CE CC 6.C00UE-01 1.1000E 01 1.9964E 01 9.3577E 00 1.1699E 01 2.8362E 00 1.0368E 01 2.6508E 00 4.1916E C1 T / T / -IT = 29 wf = 3.0079E C7 GOS = 2.00C3E 00 2.2657E-01 7.7186E-03 4.8139E-03 5.4034E-03 4.6902E-01 1.2568E-02 3.9430E-01 8.4747E-02 2.7156E-01 7.3731E-02 2.59(1E-01 1.5088E-01 E(I) = 2.6000E 01 2.500CE CC 6.CUCUE-C1 1.1000E 01 2.0095E 01 9.4386E 00 1.1835E 01 2.8303E 00 1.0366E 01 2.5787E 00 4.1251E C1 1(1)= wf = 3.0046E 07 DGS = 2.00C3E 00 2.3554E-(1 7.7829E-03 4.8515E-03 5.4456E-03 4.7186E-01 1.2661E-02 3.9623E-01 8.7864E-02 2.7105E-01 7.4440E-(2 2.8784E-01 1.4475E-01 C(I)= 2.6000E 01 2.5000E C0 6.0000E-01 1.1000E C1 2.0230E 01 9.5233E 00 1.1984E 01 2.8253E 00 1.0357E 01 1(1)= 2.5011E 00 4.055CE C1 17 = 21 NI = 3-0014F C7 005 = 2.00(3F 00 2.442tE-C1 7.6489E-03 4.8903E-03 5.4891E-03 4.7486E-01 1.2756E-02 3.9839E-01 9.1257E-02 2.7042E-01 7.5171E-02 2.7652E-C1 1.3844E-01 C(1)= 2.60CCE C1 2.50CCE CC 6.CCCCE-01 1.10COE 01 2.0371E 01 9.6125E 00 1.2146E 01 2.8212E 00 1.0342E 01 2.4173E 00 3.9936E C1 1(1)= IT = 22 MT = 2.5585E 07 DGS = 2.0063 = 002.5270E-01 7.\$1\$5E-03 4.9324E-03 5.5363E-03 4.7822E-01 1.2858E-02 4.0072E-01 9.4946E-02 2.6957E-01 7.5857E-02 2.65C3E-01 1.3159E-01 C(I)= 2.6000E 01 2.500CE 00 6.0000E-01 1.1000E 01 2.0519E 01 9.7056E 00 1.2323E 01 2.8175E 00 1.0321E 01 7.3278E 0C 3.5286E C1 401# UNITCS. EOF. REC= 00000 FIL=

CONCLUDING REMARKS

An optimization procedure for minimizing radiation shield weight has been described. The procedure, built around the steepest-descent code, OPEX-II, depends strongly on the validity of the dose-thickness relation. It has been observed that when care has been taken to accurately fit the dose-thickness relation, predictions of minimum weight configurations are quite good as verified by a necessary detailed proof calculation.

Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, January 8, 1969, 124-09-11-01-22.

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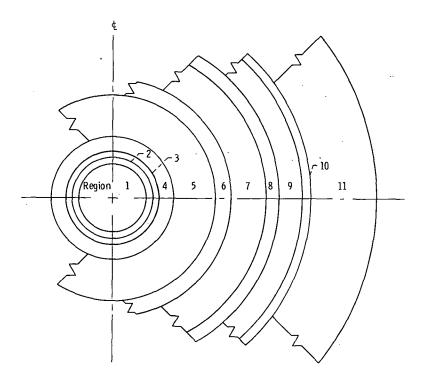
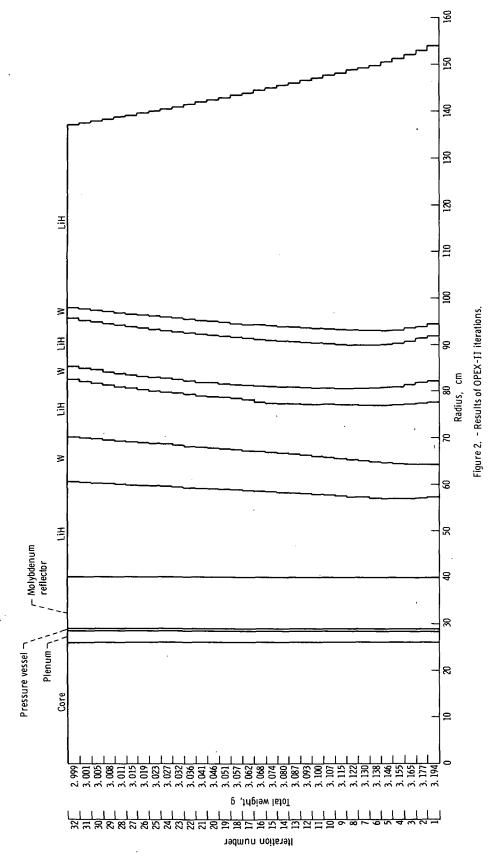


Figure 1, - Geometry for sample problem.



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